

WoodWorks

Cantilever Wood Diaphragm Webinar Series

A Design Example of a Wood Cantilever Diaphragm

Part 3- Cantilever Diaphragm Design, And Flexibility and Drift Checks



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Questions related to specific materials, methods, and services will be addressed at the conclusion of this presentation.



Course Description

Part 3 of this series will take an in-depth look at cantilever diaphragm design processes. Using calculation steps per ASCE 7 and the 2015 edition of the American Wood Council's Special Design Provisions for Wind & Seismic (SDPWS), diaphragm force distribution will be used to illustrate diaphragm design procedures in an open front structure. Diaphragm chord forces, chord slip, and chord design and detailing will all be examined. Diaphragm flexibility will also be analyzed, reviewing options proposed for the 2021 SDPWS and other rational, engineering judgement-based approaches. Finally, diaphragm deflection and story drift calculations will be explained, emphasizing factors that contribute to story drift in a cantilever diaphragm, and measurement criteria.

Learning Objectives

- 1. Discuss evolutions in mid-rise building typology that have led to the need for open-front diaphragm analysis.
- 2. Review diaphragm flexibility provisions in ASCE 7 and the 2015 Special Design Provisions for Wind & Seismic (SDPWS).
- 3. Explore one option for open-front diaphragm analysis under seismic and wind loading in a wood-frame structure.
- 4. Highlight how to calculate story drift, diaphragm deflection and torsional irregularities, and discover their effects on load distribution through a cantilever diaphragm structure.

Fasten Your Seatbelts



5 out of 5 Calculators

WoodWorks Example and Method of Analysis:

- The solutions paper and this webinar were developed independently from the AWC task group for open-front diaphragms. The method of analysis used in this example is based on our engineering judgement, experience, and interpretation of codes and standards as to how they might relate to open-front structures.
- The analysis techniques provided in this presentation are intended to demonstrate one method of analysis, but not the only means of analysis. The techniques and examples shown here are provided as guidance and information for designers and engineers.

Cantilever Wood Diaphragm Webinar Series-Content

Webinar Part 1- Code Requirements and Relative Stiffness issues:

- Introduction
- Questions needing resolution
- Horizontal distribution of shear and stiffness issues
- 2015 SDPWS open-front requirements
- Review preliminary design assumptions

Webinar Part 2- Shear Wall Design in Cantilever Diaphragm Structures:

- Introduction to open-front example
- Calculation of seismic forces and distribution
- Preliminary shear wall design
- Nominal shear wall stiffness
- Verification of shear wall design

Webinar Part 3- Cantilever Diaphragm Design, Flexibility and Drift Checks :

- Diaphragm design
- Maximum diaphragm chord force
- Diaphragm flexibility
- Story drift

Webinar Part 4-Torsional Irregularity, Other Design Checks, and Final Comments :

- Amplification of accidental torsion
- Redundancy
- Transverse direction design
- Multi-story shear wall effects

Contents and Learning Objectives

Diaphragm Design, Flexibility Check, and Drift:

Diaphragm design

Calculate the maximum diaphragm design force and learn various ways to distribute torsional forces into a diaphragm.

Maximum diaphragm chord force

Discover the maximum chord force if shear walls are located along the chord line or concentrated in-plane wall loads are located at the ends of a cantilever.

Diaphragm flexibility

Cantilever diaphragm deflection equations will be presented. Learn how to determine cantilever diaphragm flexibility.

Story drift

The method of calculating story drift for a cantilever diaphragm under seismic or wind loading will be presented. Options for reducing story drift or torsional irregularities will be provided if allowable drift is exceeded.

Diaphragm Design



Diaphragm Design Forces: MSFRS or Fpx



Page 28

Questions

- 1. When does a loss in stiffness in the exterior walls cause an open-front diaphragm condition?
- 2. What is the deflection equation for open-front/cantilever diaphragms?
- 3. How is diaphragm flexibility defined for open-front/cantilever diaphragms vs. ASCE 7-16, Figure 12.3-1?
- 4. What are the available methods of distributing torsional forces into the diaphragm?
- 5. Do shear walls located along diaphragm chord lines affect the diaphragm chord forces?
- 6. Will the in-plane lateral forces of the exterior walls located at the ends of the cantilever increase chord forces, or is it acceptable to include these as part of the PSF lateral load?
- 7. How are torsional irregularities determined and addressed for openfront/cantilever diaphragms?

12.10.1.1 Diaphragm Design Forces.

The diaphragm must be designed to the maximum of these two:

- MSFRS Diaphragm (structure) Load (F_x) or,
- Controlling Diaphragm inertial Design Load (F_{px}) Per Eq. 12.10-1 as follows:

$$\begin{split} F_{px} &= \frac{\sum_{i=x}^{n} F_{i}}{\sum_{i=x}^{n} w_{i}} w_{px} \end{split} \tag{12.10-1} \\ & \text{where} \\ F_{px} &= \text{the diaphragm design force at level } x \\ F_{i} &= \text{the design force applied to level } i \\ w_{i} &= \text{the weight tributary to level } i \\ w_{px} &= \text{the weight tributary to the diaphragm at level } x \end{split}$$

The force shall not be less than $F_{px} = 0.2S_{DS} lew_{px}$ (12.10-2)

The force need not exceed $F_{px} = 0.4S_{DS} lew_{px}$ (12.10-3)

For inertial forces calculated in accordance with Eq. 12.10-1, $\rho=1.0$ per ASCE 7-16 Section 12.3.4.1, Item 7.

For a single story structure $F_x = F_{px} = \frac{S_{DS}I_e}{R}w_{px}$



Using method 2B- ρ=1.3, Ax=1.25 :

 F_T = Torsion forces only at corridor walls, gridlines 2 and 3 M_{net} = 480.1(6 ft.) = 2880.6 ft. lbs. Net moment

The in-plane forces of the longitudinal walls applied at grid lines 1, 2, 3 and 4 are calculated:

F1,2,3,4= 0. 167(0.7)(1.3)(13 psf)
$$\left(\frac{10}{2} + 2\right)$$
(40) = 553. 2 lbs.

Vnet= Vbase- F1,2,3,4 =12438.3(1.3) - 4(553.2) = 13957 lbs.

$$W = \frac{13957}{76} = 183.65$$
 plf uniform load



Corridor walls

Mnet

 $W_T = \frac{2880.6}{38(38)} = 2.0$ plf: equivalent uniform torsional load acting as Mnet

W1 = 183.65 – 2.0 = 181.65 plf: uniform load minus torsional load=net uniform load left cantilever Wall Load







Diaphragm Capacity-Wood Structural Panels

Table 4.2A Nominal Unit Shear Capacities for Wood-Framed Diaphragms^{1,3,6,7}

Blocked

							Α			E	3	
							Seismic			Wi	nd	
Sheathing	Common	Minimum	Minimum	Minimum	Nail spaci	ng (in.) at bou	undaries (all ca	ases), at	Dan	ol Eda	o Easta	nor
Grade	nail Size	Fastener	Nominal	Nominal width	continuous p	oanel edges p	(cases 3 &	Fair	Cr Lug			
		Penetration	Panel	Of nailed face	4), and	<u>l at all panel e</u>	<u>& 6).</u>		Spaci	ig (iii.)	/	
		In Framing	Thickness	At adjoining	6	4	2 1/2	2	6	4	2 ½	2
		Member or	(in.)	Panel edges	Nail spacing (i	n.) at other p	anel edges(cas	ses 1, 2, 3 & 4				
		Blocking	, ,	and boundaries	6	6	4	3	6	6	4	3
		(in.)		(in.)	Vs Ga	Vs Ga	Vs Ga	Vs Ga	Vw	Vw	Vw	Vw
					(plf) (kips/in.	(plf) (kips/in.	(plf) (kips/in.	plf) (kips/in.) (plf)	(plf)	(plf)	(plf)
					OSB PLY	OSB PLY	OSB PLY	OSB PLY				

	04	1 2/0	7/16	3	570	11	9	760	7	6	1140	10	8	1290	17 12	800	1065	1595	1805
Chaothing	ou	1-3/0	15/22	2	540	13	9.5	720	7.5	6.5	1060	11	8.5	1200	19 13	755	1010	1485	1680
Sneathing			15/52	3	600	10	8.5	800	6	5.5	1200	9	7.5	1350	15 11	840	1120	1680	1890
and Circula floor			15/22	2	580	25	15	770	15	11	1150	21	14	1310	33 18	810	1080	1610	1835
Single floor	101		15/52	3	650	21	14	860	12	9.5	1300	17	12	1470	28 16	910	1205	1820	2060
	10d	1-1/2	10/22	2	640	21	14	850	13	9.5	1280	18	12	1460	28 17	895	1190	1790	2045
			19/32	3	720	17	12	960	10	8	1440	14	11	1640	24 15	1010	1345	2015	2295

Roof framing-D.F. 1, E = 1,700,000 psi, roof joists @ 16" 0.c.

Unit torsional shear = 24.32 plf

 $V_{Max diaph} = 176.3 + 24.3 = 200.6 \text{ plf.}$

```
200.6 plf < 0.5(580) = 290 plf. o.k.
```

Ga = 25, blocked





By inspection, the walls along the chord line affect the chord forces by a small amount, 364.8 lbs.
 Calculations show that the conc. wall force at end of cantilever increase the chord force by +21% at

the 15'splice diminishing to +9% increase at 23', and +1% at the support. Walls had a larger effect.

Diaphragm Chords

	Diaphr	agm De	flection	(ASD)									
	Spli	ce Forces (L	bs.)	Σδ_slip	v unif.	v conc.	Ga	L'	W'	δDiaph Uni	iðDiaph con	Total δ	
	F 15	F23	F35	In.	plf	plf	k/in.	Ft.	Ft.	In.	In.	11.	
	1094.3	1180.9	3253.7	0.072	186.75	13.83	25.0	35.00	40.00	0.225	0.02	0.248	Rt. Cantilever
Nails Req'd=	4.84	5.23	14.40					ord ce rd ce	2 0				
Use Nails =	8	16	24		Wall Load			pli pli					
Slip=	0.023	0.013	0.023		553.2		553.2	553.2	· O.ø	553.2			
	EA= 28050	000, (2)2x6							185.64				
	lincludes e	ffects of sw	's along ch	ord line		181.65							
					<u>t</u> <u>t</u> <u>t</u>	*****	• • • • • • •	<u>++ +</u> + + +		<u>+ + + +</u>		W2	W1
					Μ	ethod 2B	T T					183.65	183.65
												2.0	-2.0
							7604.8	8565.0				185.64	181.65
	Diaphr	agm De	flection	(ASD)								\frown	
	353.6	1884.0	3338.5	0.070	183.26	13.83	25.0	35.00	40.00	0.219	0.02	0.243	Lft. Cantilever
	1.56	8.34	14.77										
	8	16	24										
	0.008	0.020	0.024										

Maximum chord force = 3338.5 lbs.

Using (2)2x6 DF-Larch No.1 wall top plates as the diaphragm chords: 2015 NDS Supplement Table 4A Ft = 675 psi, Fc = 1500 psi. Only one 2x6 plate resists the chord forces due to the nailed splice joint.

$$f_t = \frac{F_{chord}}{(1)2x6}$$
, Number of nails = $\frac{F_{chord}}{226}$, where 226 lbs. is adjusted lateral design value, Z' (ASD), for 16d nails (face nailed).

Compression stresses OK by inspection. Chords braced about both axes.

Check for Effects of Full Length Shear Walls on Chord Forces







Seismic- ρ=1.0, Ax=1.25

Page 41

• ASCE 7-16 Diaphragm Flexibility

- 12.3.1.1 Flexible Diaphragm Condition.
 - Untopped steel decking or wood structural panels
 - Permitted to be <u>idealized as flexible</u> under certain conditions.
- 12.3.1.2 Rigid Diaphragm Condition.
 - Concrete slabs or concrete-filled metal deck (No mention of wood)
 - Span-to-depth ratios of 3 or less with no horizontal irregularities
 - Permitted to be <u>idealized as rigid.</u>
- 12.3.1.3 Calculated Flexible Diaphragm Condition. (No calculated Rigid condition)
 - Diaphragms not satisfying the conditions of Sections 12.3.1.1or 12.3.1.2
 - Permitted to be <u>idealized as flexible</u> provided: $\delta MDD > 2\Delta ADVE$.
- 2018 IBC Section 1604.4:
 - O A diaphragm is <u>rigid</u> when δMDD ≤ 2ΔADVE.



Determination of Cantilever Diaphragm Flexibility (Question 3):

Page 42



(b) Corridor Walls Only Preferred Method Can require engineering judgement

ATC WDSC -Proposed language

Cantilevered diaphragms shall be permitted to be idealized as rigid when the calculated maximum inplane deflection of the diaphragm itself under lateral load is less than or equal to two times the deflection of vertical elements of the lateral force-resisting system of the associated story used to determine the cantilever length, L' (See Figure 4A).



Cantilever Diaphragm Deflection Equations (Question 2): AWC WDSC Ballot 4

Three-term equation for uniform load:

$$\delta_{Diaph \, Unif} = \frac{3\nu L'^3}{EAW'} + \frac{0.5\nu L'}{1000G_a} + \frac{\Sigma x'\Delta_C}{W'}$$

Four-term equation for uniform load:

$$\delta_{Diaph Unif} = \frac{3\nu L'^3}{EAW'} + \frac{0.5\nu L'}{G\nu t\nu} + 0.376 L' e_n + \frac{\Sigma x'\Delta_C}{W'}$$

Three-term equation for point load:

$$\delta_{Diaph \ Conc} = \frac{8\nu L'^3}{EAW'} + \frac{\nu L'}{1000G_a} + \frac{\Sigma \mathbf{x}' \Delta_C}{W'}$$

Four-term equation for point load:

$$\delta_{Diaph \ Conc} = \frac{8\nu L'^3}{EAW'} + \frac{\nu L'}{G\nu t\nu} + 0.75 \ L' \ e_n + \frac{\Sigma x' \Delta_C}{W'}$$

For method 2B, the maximum diaphragm deflection is equal to the sum of the uniform load deflection plus the concentrated load deflection:

EA chords =28,050,000 lbs., 2-2x6 wall top plate.

Where:

- L' = cantilever diaphragm length, ft
- W' = cantilever diaphragm width, ft
- E = modulus of elasticity of diaphragm chords, psi
- A = area of chord cross-section, in.2
- v_{max} = induced unit shear at the support from a uniform applied load, lbs/ft
- G_a = apparent diaphragm shear stiffness from nail slip and panel shear deformation, kips/in
- *Gvtv* = Panel rigidity through the thickness
- X' = distance from chord splice to the free edge of the diaphragm, ft
- $\Delta_{\rm c}$ = diaphragm chord splice slip, in.
- $\delta_{Diaph Unif}$ = calculated deflection at the free edge of the diaphragm, in.
- e_n Nail slip per SDPWS C4.2.2D for the load per fastener at v_{max}

 $\delta_{Diaph \ Conc}$ = calculated deflection at the free edge of the diaphragm, in.



Page 39

Longitue	dinal Lo	ading e=	=4.75', T =	84403 ft.	lbs., <mark>ρ=1.0</mark>	, Ax=1.25				
Grid Line	kx	Ку	dx	dy	kd	kd ²	Fv	Fт	Fv+FT	b B
2	43.54		3		130.63	391.89	8884.5	-527.7	8356.8	rric
3	43.54		3		130.63	391.89	8884.5	527.7	9412.2	ီပိ
Α		25.14		20	502.74	10054.73		2030.9	2030.9	ם.
В		25.14		20	502.74	10054.73		-2030.9	-2030.9	ק פ
Σ	87.09	50.27			J=	20893.23	17769			at
				-						alls
										Ň

	Diaphr	agm De	eflection	(STR)								Rt. Cantilever
	Spli	ce Forces (L	bs.)	Σδ_slip	v unif.	v conc.	Ga	L'	W'	δDiaph Unif	5Diaph con	Total δ
	F 15	F23	F35	In.	plf	plf	k/in.	Ft.	Ft.	In.	In.	In.
	1064.6	1159.7	3533.5	0.075	233.22	0.00	25.0	35.00	40.00	0.265	0.00	0.265
ails Req'd=	4.71	5.13	15.64			ρ _Φ	0 0					
Use Nails =	8	16	24			lice	lice		olia clia			
Slip=	0.023	0.012	0.025			ې مې م	τ <mark>ο</mark> α		<u>o.</u> o. o. o			
	EA= 28050	000, (2)2x6							236.00			
	lincludes e	ffects of sw	's along ch	ord line		231.61						
					<u> </u>	· ★ ★ ★ <mark></mark> ★ ★ `			<u>+ + + +</u> +			
					Μ	ethod 2A		1				
							8356.8	9412.2				
	Diaphr	agm De	flection	(STR)								Lft. Cantilever
	250.6	1932.4	3626.7	0.073	229.38	0.00	25.0	35.00	40.00	0.260	0.00	0.260
	1.11	8.55	16.05									
	8	16	24									
	0.005	0.021	0.026									

Flexibility and Drift Page 43

Diaphragm Deflection-Method 2A, p=1.0, Ax=1.25

 $\delta_{Diaph \, Unif} = \frac{3v_{max}L'^3}{EAW'} + \frac{0.5v_{max}L'}{1000G_a} + \frac{\Sigma A_C X_C}{W'}$ Three-term equation for uniform load

Wall displacements from Spreadsheet:

 $\delta_{Diaph \ left} = 0.26$ ", $\delta_{Diaph \ right} = 0.265$ "

Deflection at grid line 3 = 0.216"

 $2 \operatorname{x} \Delta_3 = 0.432^{"}$

0.265" < 0.432" ∴ Diaphragm can be idealized as Rigid

Diaphragm Flexibility – Wind

- ASCE 7-16, Chapter 27, Section 27.5.4-DIAPHRAGM FLEXIBILITY-requires that the structural analysis <u>shall</u> consider the stiffness of diaphragms and vertical elements of the main wind force resisting system (MWFRS).
- Section 26.2 Definitions, DIAPHRAGM, diaphragms constructed of WSP are permitted to be idealized as flexible.
- ASCE 7 drift limits requirements for wind design can be found in ASCE 7-16 Commentary Appendix CC, Serviceability Considerations Section CC2.2 Drift of Walls and Frames





Legend

- Engineering judgement required
- → SW & Diaph. Design
- **Determine flexibility, Drift**

→ Determine Tors. Irreg., ρ, Ax

ASD Design

STR Design



Check Story Drift Seismic- p=1.0, Ax=1.25 Page 44



ASCE 7-16 Section 12.8.6-Story Drift Determination Regular structures:

- Story drift (Δ) shall be computed as the difference of the deflections <u>at the centers of</u> <u>mass</u> at the top and bottom of the story under consideration (Fig. 12.8-2).
- For structures assigned to SDC C, D, E, or F that have <u>horizontal irregularity Type 1a or 1b</u> of Table 12.3-1, the design story drift, Δ , shall be computed as the largest difference of the deflections of vertically aligned points at the top and bottom of the story under consideration <u>along any of the edges</u> of the structure.

SDPWS Section 4.2.5.2 (4): Open-front structures, loading parallel to the open side:

Maximum story drift <u>at each edge</u> of the structure ≤ ASCE 7-16 allowable story drift (Seismic) including torsion and accidental torsion and shall include shear and bending deformations of the diaphragm computed - strength level basis amplified by C_d.

$$\delta_{\rm X} = \frac{C_d \delta_{xe}}{I_e}$$



Drift-Method 2A (Seismic) p=1.0, Ax=1.25

Drift
$$\Delta = \delta_{Diaph} + \delta_{Rotation} + \delta_{Translation}$$

 $\delta_2 = 8.357 \text{ k} / 43.54 \text{ k/in} = 0.192 \text{ in},$
 $\delta_3 = 9.412 \text{ k} / 43.54 \text{ k/in} = 0.216 \text{ in}$
 $\delta_A = 2.031 \text{ k} / 25.14 \text{ k/in} = 0.081 \text{ in},$
 $\delta_B = -2.031 \text{ k} / 25.14 \text{ k/in} = -0.081 \text{ in}$
 $\Delta_{Diaph} = 0.265^{\circ}$
 $\Delta_{Average} = 0.204^{\circ}$ (Translation)
 $\delta_{RL} = \frac{2\Delta_{SWA,B}(L'+3')}{W'} = \frac{2(0.081)(35'+3')}{40} = 0.154^{\circ}$

Drift
$$\Delta = \sqrt{(\delta_T + \delta_D \pm \delta_{RL})^2 + (\delta_{RT})^2}$$

Drift
$$\Delta_4 = \sqrt{(0.204 + 0.265 + 0.154)^2 + (0.081)^2} = 0.628''$$

Drift
$$\Delta_1 = \sqrt{(0.204 + 0.26 - 0.154)^2 + (0.081)^2} = 0.320$$
"

Cd = 4, le = 1

$$\delta_M = \frac{C_d \delta_{max}}{I_e} = \frac{4(0.628)}{1} = 2.51"$$



 $\delta_{RT} = 0.081''$

- $$\begin{split} \delta_{RT} &= Transverse \ component \\ of \ rotation \\ \delta_{RL} &= Longitudinal \ component \end{split}$$
- of rotation
- δ_D =Diaphragm displacement
- δ_T = Translational displacement

Table 12.12-1 Allowable Story	/ Drift, Δa	l	
		Risk Categ	jory
Structure	l or ll	III	IV
Structures, other than masonry shear wall structures, four stories or less above the base as defined in Section 11.2, with interior walls, partitions, ceilings, and exterior wall systems that have been designed to accommodate the story drifts.	0.025hsx	0.020hsx	0.015hsx
Masonry cantilever shear wall structures	0.010hsx	0.010hsx	0.010hsx
Other masonry shear wall structures	0.007hsx	0.007hsx	0.007hsx
All other structures	0.020hsx	0.015hsx	0.010hsx

- Depends on the non-structural components and detailing.
- Most sheathed wood framed walls can undergo the 2.5% drift level while providing life safety performance at the seismic design level.
- 0.025hsx limit interior walls, partitions, ceilings, and exterior walls can accommodate the higher story drift limit. The selection of the higher 2.5% drift limit should be taken only with consideration of the non-structural wall and window performance.
- Otherwise, the 2% drift limit requirements should be used.

 $0.025hsx = 0.025(10)(12) = 3.0" > 2.51" \therefore drift O.K.$

0.02hsx = 0.02(10)(12) = 2.4" < 2.51" ∴ drift not O.K. for 2% drift Page 47

Solutions if drift is exceeded: Page 48

Additional stiffness must be provided in either the diaphragm or in the shear walls:

- a. Diaphragms-
 - Increasing nail size, spacing and/or sheathing thickness can increase shear capacity but <u>it will not, in most cases</u>, increase the diaphragm stiffness, if using the 3 term eq.
 - The largest deflection comes from shear deflection and nail slip.
 - SDPWS Table 4.2A shows that the apparent shear stiffness diminishes as you decrease the boundary nail spacing from a 6/6/12 nailing pattern until you get to a 2/3/12 nailing pattern.
 - If using plywood, switch to OSB which has a higher Ga

Table 4.2A Nominal Unit Shear Capacities for Wood-Framed Diaphragms Blocked

										4					E	3	
									Seis	smic					Wi	nd	
Sheathing	Common	Minimum	Minimum	Minimum		Nail s	paci	ing (in.) at bo	unda	ries (al	cases	s), at		Dor		To Fact	onor
Grade	nail Size	Fastener	Nominal	Nominal width	со	ntinuo	ous	panel edges p	paralle	el to lo	ad (ca	ses 3 8	&	Par	el Eug	e rasu	ener
		Penetration	Panel	Of nailed face		4),	_		Space	ng (in.))						
		In Framing	Thickness	At adioining		6		4		2 ½		2		6	4	2 ½	2
		Member or	(in.)	Panel edges	Nail	spaciı	rg (i	n.) at <mark>other p</mark>	anel	dges(ases	l, 2, 3	& 4	I)			
		Blocking		and boundaries		6		6		4		3		6	6	4	3
		(in.)		(in.)	V	s G	a	Vs Ga	Vs	Ga	Vs	s Ga	3	Vw	Vw	Vw	Vw
					(plf	<u>) (kips</u>	/in.	(plf) (kips/in	.)(plf)	(kips/i	n.plf) (kips,	/in) (plf)	(plf)	(plf)	(plf)
						OSB	PLY	OSB PLY		OSB P	LY	OSB P	PLY				

	ام ۵	1 2 / 0	7/16	3	570	11	9	760	7	6	1140	10	8	1290	17 12	800	1065	1595	1805
a	δü	1-3/0	15/22	2	540	13	9.5	720	7.5	6.5	1060	11	8.5	L200	19 13	755	1010	1485	1680
Sheathing			15/32	3	600	10	8.5	800	6	5.5	1200	9	7.5	1350	15 11	840	1120	1680	1890
and			15/22	2	580	25	15	770	15	11	1150	21	14	1310	<mark>33</mark> 18	810	1080	1610	1835
Single floor	101		15/52	3	650	21	14	860	12	9.5	1300	17	12	1470	28 16	910	1205	1820	2060
	10d	1-1/2	10/22	2	640	21	14	850	13	9.5	1280	18	12	1460	28 17	895	1190	1790	2045
			19/32	3	720	17	12	960	10	8	1440	14	11	1640	24 15	1010	1345	2015	2295

- b. <u>Shear walls</u>- Contrary to the diaphragm, decreasing the nail spacing on the shear walls would increase the wall stiffness, reference SDPWS Table 4.3A. The apparent shear stiffness, Ga, increases as the nail spacing decreases.
- c. <u>Other options to increase stiffness:</u>
 - Increase the wall lengths.
 - Increase the number of shear walls in the lateral line of force-resistance.
 - Apply sheathing to both sides of the walls at grid lines A & B or decrease nail spacing.
 - Decrease nail spacing at corridor walls.
 - Increase the size of the hold downs(with smaller ∆a) to lessen rod elongation and wall rotation.
 - Increase the number of boundary studs (decrease bearing perpendicular to grain stresses, crushing). Optional LVL or LSL plates.
 - Add additional interior shear walls to decrease forces on other shear walls.

d. Calculation Method: A final option which may increase the calculated system stiffness and reduce the deflections is to use the four-term deflection equation for the shear wall and diaphragm deflections to avoid introducing an artificial bias in the results by selectively combining three-term and four-term deflection calculations.

Solution for 2% drift issue: Page 50

Following option (d), the 2% drift limit can potentially be achieved by using the four-term deflection equation, which reduces diaphragm deflection and drift, as noted below.

$$\delta_{Diaph \, Unif} = \frac{3vL'^3}{EAW'} + \frac{0.5vL'}{Gvtv} + 0.376 \ L' \ e_n + \frac{\Sigma x \Delta_C}{W'}$$

Where:

$$e_n = \left(\frac{V_n}{769}\right)^{3.276} = \left(\frac{116.6}{769}\right)^{3.276} = 0.002 in$$
 SDPWS Table C4.2.2D

where 116.6 is max. load per nail, 10d nails, dry lumber assumed.

SDPWS Table C4.2.2A

v = 233.2 plf

 $\frac{2\Sigma x \Delta_c}{W'} = \frac{2[15(0.023) + 23(0.012) + 35(0.025)]}{40} = 0.075 \text{ in}$ $\delta_{\text{Diaph Unif}} = \frac{3(233.2)35^3}{28050000(40)} + \frac{0.5(233.2)35}{35000} + 0.376(35)0.002 + 0.075 = 0.245 \text{ in}$ Drift $\Delta_4 = \sqrt{(0.204 + 0.245 + 0.153)^2 + (0.081)^2} = 0.608 \text{ in}$

 $\delta_{\rm M} = \frac{C_d \delta_{max}}{I_e} = \frac{4(0.608)}{1} = 2.434$ in. ≈ 2.4 in. Close enough to comply with the 2% drift limitation. Drift can also be improved if ρ or Ax decreases (See Section 7.6.1).

Check for Wind Drift Simplified Procedure Chapter 28, Part 1 Low-rise Buildings, Enclosed

ASCE 7-16 Section 2.4 ASD LC 0.6D+0.6W



Kd=0.85	Wind directionality factor	26.8
GCpi=+/-0.18 ₍₂₎	Internal pressure coeff.	26.13
$Kz=2.01\left(\frac{15}{z_g}\right)^{(\alpha)}$	Velocity pressure exp. coeff.	26.10-1
Kz=0.78 @ h=10'		
$Qh=0.00256K_ZK_{ZT}$	$K_d V^2 = 22.4 \text{ psf}$	26.10-1

Figure 28.3-1

	Surface	1	4	1E	4E										
	GCpi	0.4	-0.29	0.61	-0.43										
	P (psf)	8.96	6.5	13.66	9.63										
	<u>Parapet</u> Pp=Qp(GC Kz=0.85 @	15.4 ^(pn) 12' Top o	46 psf of parapet	23.	3 psf 28.3	3-2									
	Qp=24.46 psf GCpn ww=1.5, GCpn lw=-1.0 28.3.2 Ppw=36.69 psf, Ppl=24.46 psf ∑Pp=61.15 psf														
K			Para 2 4E	apet ,3 1,4											
		22	=8'												

Rigid 1	<u>Diaphr</u>	agm A	nalysis	<u>(ASD</u>)		Mind	1 \ /	110						Requires Input			
Longitud	linal Loa	nding					VVIIIC	Vult–	.112	IVIPI		s						
Grid Line	kx	Ку	dx	dy	kd	kd ²	Fv		Fт	Fv+FT	Loads	^o SW	Rho=	1		2a=	8	ļ
2	43.54		3		130.63	391.89	4923.8		-40.0	4883.8		0.112	Ax=	1		Net=	23.5]
3	43.54		3		130.63	391.89	4923.8		40.0	4963.8		0.114						
A		25.14		20	502.74	10054.72756			153.8	153.8	+	0.0061	Fy=	9847.6		W1,4=	127.1	
В		25.14		20	502.74	10054.72756			-153.8	-153.8		-0.006	e=	34		W1E,4E=	150.6	Į
Σ	87.09	50.27			J=	20893.23102	9847.6							<i>T</i> =	6392.0			
Transver	rse Load	ing																
Grid Line	kx	Ку	dx	dy	kd	kd ²	Fv		Fт	Fv+FT						Shear wa	all ρ=1.3,	Ax=1.25
2	43.54		3		130.63	391.89			18.8	18.8		0.000				Torsion,	Αx ρ=1.0	, Ax=1.0
3	43.54		3		130.63	391.89			-18.8	-18.8	Loads	0.000				Flex/Drif	ft ρ=1.0 <i>, l</i>	4x=1.25
Α		25.14		20	502.74	10054.72756	4923.8		72.4	4996.2		0.199	Fx=	9847.6		Redunda	ancy ρ=1.	0, Ax=1.0
В		25.14		20	502.74	10054.72756	4923.8		-72.4	4851.4		0.193	emin=	16				
Σ	87.09	50.27			J=	20893.23102	9847.6							<i>T</i> =	3008.0			

Use this load	combina	tion for d	lefining N	Nominal S	tiffness v	alues, Keff.	Then use t	those Kei	ff values for	r all other a	analyses.		5-11=-	in in	na,		
Expected I	Dead + S	Seismic	D+QE	(other ter	ms if "ex	pected" gra	vity loads a	as per AS	ρ=1.0 <i>,</i> Ax=	1.0			and F	10 1000	i. b		
Grid Line	SW	Ga	Rho	V on wall	v	Т	С	Δ_a	$F_{c\perp}$	Crush.	Shrink	δ 	δ s	δ _{Rot}	δ sw		K (k/in)
Calculate S	Stiffness of \	Nalls on A 8	& B using T	ransverse lo	ading												
Α		37	1.0	7308.0	913.5	6390.8	13770	0.154	556.36	0.056	0.019	0.022	0.247	0.313	0.581	Α	25.14
В		37	1.0	7308.0	913.5	6390.8	13770	0.154	556.36	0.056	0.019	0.022	0.247	0.313	0.581	В	25.14
Calculate S	tiffness of \	Nalls on 2 8	& 3 using <mark>Lo</mark>	ongitudinal	loading												25.14
2		30	1.0	7022.0	702.2	6391.1	8340.7	0.154	505.50	0.045	0.019	0.020	0.234	0.230	0.484	2	43.54
3		30	1.0	7022.0	702.2	6391.1	8340.7	0.154	505.50	0.045	0.019	0.020	0.234	0.230	0.484	3	43.54
		V equa	al to revise	d wall force	based on H	D STR (design)	capacity		625 Max.	Add stud							43.54

						6	3	vL 3 0.5	vmart"	EAcXc			
Diaphragm Deflection (STR) $\delta_{Diaph Unif} = \frac{\delta_{Diaph Unif}}{EAW} + \frac{\delta_{Diaph Unif}}{1000G_a} + \frac{\delta_{W}}{W}$ Rt. Cantilever													
Splice Forces (Lbs.)			Σδ_slip	v unif.	v conc.	Ga	L'	W'	δDiaph Un	if SDiaph con	Total δ		D
F 15	F23	F35	In.	plf	plf	k/in.	Ft.	Ft.	In.	In.	In.		P
395.2	827.5	1980.6	0.041	115.55	0.00	25.0	35.00	40.00	0.135	0.00	0.135		
1.75	3.66	8.76						2020					
8	16	24						a in cina			lent]!!!!!!!!!!
0.008	0.009	0.014				1		0 0 0.0		equiva			
EA= 28050000, (2)2x6					i		129.74	_					
lincludes effects of sw's along chord line					129.41	!							-24'
				<u><u> </u></u>	<u>· ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓ ↓</u>	<u> </u>	** * * *				W2	W1	€-34
				Method 2A			1				129.57	129.57	
											0.17	-0.17	
						4883.8	4963.8				129.74	129.41	
Diaph	ragm D	Deflection	on (STI	R)							L	ft. Cantileve	er
333.6	886.1	1987.6	0.041	115.26	0.00	25.0	0 3	5.00 4	0.00	0.135	0.00	0.135	
1.48	3.92	8.79											
8	16	24											
0.007	0.009	0.014											

Wind Design (ASD) Drift-Similar to Method 2A 0.6D+0.6W



Drift-Method 2A (Wind) p=1.0, Ax=1.25

Drift
$$\Delta = \delta_{Diaph} + \delta_{Rotation} + \delta_{Translation}$$

 $\delta_2 = 4.884 \text{ k} / 43.54 \text{ k/in} = 0.112 \text{ in},$
 $\delta_3 = 4.964 \text{ k} / 43.54 \text{ k/in} = 0.114 \text{ in}$
 $\delta_A = 0.154 \text{ k} / 25.14 \text{ k/in} = 0.0061 \text{ in},$
 $\delta_B = -0.154 \text{ k} / 25.14 \text{ k/in} = -0.0061 \text{ in}$
 $\Delta_{Diaph} = 0.135^{\circ}$
 $\Delta_{Average} = 0.113^{\circ} \text{ (Translation)}$
 $\delta_{RL} = \frac{2\Delta_{SWA,B}(L'+3')}{W'} = \frac{2(0.0061)(35'+3')}{40} = 0.0116^{\circ}, \delta_{RT} = 0.0061$
Drift $\Delta = \sqrt{(\delta_T + \delta_D \pm \delta_{RL})^2 + (\delta_{RT})^2}$
Drift $\Delta_4 = \sqrt{(0.113 + 0.135 + 0.0116)^2 + (0.0061)^2} = 0.26^{\circ}$
Drift $\Delta_1 = \sqrt{(0.113 + 0.135 - 0.0116)^2 + (0.0061)^2} = 0.24^{\circ}$

Drift $\Delta_4 = 0.26$ "

Flexibility check:

 $\delta_{\text{MDD}} < 2 \ \delta_{\text{ADVE}}, \ 0.135^{"} < 2(0.113^{"}) = 0.226^{"}$ Diaphragm is rigid or semi-rigid



 δ_{RT} = Transverse component of rotation δ_{RL} = Longitudinal component

6"

- of rotation
- δ_D =Diaphragm displacement
- δ_T = Translational displacement

Allowable Drift Wind H/600, H/400



Assuming window manufacturers allowable tolerance (movement) =0.25" (Check with window manufacturer)

10' wall hgt.

H/600 =0.2" < 0.26" NG by inspection H/400 =0.3" at top of wall Drift Δ_4 =0.26"<0.3" ∴ drift OK Maximum displacement at top of window at allow doft =0.21"<0.25"

window at allow defl.=0.21"<0.25" ∴ OK

9' wall hgt.

H/400 =0.27" at top of wall 0.26"<0.27" ∴ drift OK

Maximum displacement at top of window=0.21"<0.25" ∴ OK

For resistance to Wind loads:

 ASCE 7 drift limits requirements for wind design can be found in ASCE 7-16 Commentary Appendix CC, Serviceability Considerations Section CC2.2 Drift of Walls and Frames

Damage to nonstructural partitions, cladding, and glazing may occur if the story drift exceeds about 3/8" unless special detailing practices are made to tolerate movement

Questions?

This concludes Woodworks Presentation on:

Part 3- Cantilever Diaphragm Design, And Flexibility and Drift Checks

Your comments and suggestions are valued. They <u>will</u> make a difference.

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Thank You

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