### U.S. Mass Timber Floor Vibration

#### Design Guide





# Mass Timber Floor Design for Walking-Induced Vibrations

Reid Zimmerman, PE, SE Technical Director | KPFF Portland

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## Acknowledgements

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Developed for WoodWorks - Wood Products Council by:

Scott Breneman, PhD, PE, SE, WoodWorks – Wood Products Council Reid Zimmerman, PE, SE, KPFF Consulting Engineers Adam Gerber, PEng, MASc, Aspect Structural Engineers Lucas Epp, PEng, PE, StructureCraft Carla Dickof, PEng, MASc, Fast + Epp Andrew Taylor, PhD, PE, SE, KPFF Consulting Engineers William Loasby, PEng, CEng, Fast + Epp Eric McDonnell, PE, Holmes Structures Christian Slotboom, EIT, Fast + Epp Jack McCutcheon, PE, KPFF Consulting Engineers Rene Visscher, EIT, StructureCraft

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## Outline

- Vibration Sensitivity and Subjectivity
- Understanding Floor Vibration
- Floor Vibration Criteria
- Mass Timber Floor Vibration Design Decisions
- Mass Timber Floor Vibration Design Methods and Modeling
- Mass Timber Floor Vibration Design Examples

## Vibration Sensitivity and Subjectivity

## Vibration Sensitivity and Subjectivity



## Vibration Sensitivity and Subjectivity









FIGURE 2-1: SDOF vibration

$$f_n = \frac{1}{2\pi} \sqrt{\frac{k}{m}}$$







TABLE 3-7: Example walking frequencies (AISC Design Guide 11)

Walking Speed	Walking Frequency (Hz)	Steps Per Minute (SPM)	Potential Occupancies				
Very slow (uncommon)	1.25	75	Laboratories, surgical theaters				
Slow	1.6	95	Bedrooms, hotel rooms				
Moderate	1.85	110	Residential living areas, office work areas				
Fast	2.1	126	Corridors, shopping malls, airports				

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FIGURE 2-2: Build-up to resonant response (left) and transient response (right) of damped systems

## **Floor Vibration Criteria**

## **Floor Vibration Criteria**



#### TABLE 3-13: Suggested performance targets

Place	Peak Acceleration Target	RMS Velocity Target			
Offices or residences	0.5% g	16,000-32,000 mips			
Premium offices or luxury residences	0.3% g	8,000-16,000 mips			

## **Floor Vibration Criteria**





TABLE 3-14: Generic velocity criteria for sensitive equipment

Designation	RMS Velocity Limit	Application
N/A	6,000 mips	Hospital patient rooms
N/A	4,000 mips	Surgery facilities, laboratory robots, bench microscopes up to 100x, operating rooms
VC-A	2,000 mips	Microbalances, optical comparators, mass spectrometers, industrial metrology laboratories, spectrophotometers, bench microscopes up to 400x
VC-B	1,000 mips	Microsurgery, microtomes and cyrotomes for 5-10 µm slices, tissue and cell cultures, optical equipment on isolation tables, bench microscopes greater than 400x, atomic force microscopes
VC-C	500 mips	High-precision balances, spectrophotometers, magnetic resonance imagers, microtomes and cyrotomes for < 5 μm slides, chemotaxs, electron microscopes at up to 30,000x
VC-D	250 mips	Cell implant equipment, micromanipulation confocal microscopes, high-resolutions mass spectrometers, electron microscopes at greater than 30,000x
VC-E	125 mips	Unisolated optical research systems, extraordinarily sensitive systems

### Parameters within design control



### "Free" Stiffness

- Dynamic modulus of concrete,  $E_{c,dyn} = 1.35^*E_c$
- Appropriate modeling of connections for vibration evaluation
- Nonstructural walls
- Incidental timber-timber or timber-concrete composite action





$$EI_{eff} = EI_1 + EI_2 + \gamma_1 * EA_1 * a_1^2 + \gamma_2 * EA_2 * a_2^2$$

$$a_{1} = \frac{\gamma_{2} * (EA)_{2} * r}{\gamma_{1} * (EA)_{1} + \gamma_{2} * (EA)_{2}}$$

$$a_{2} = \frac{\gamma_{1} * (EA)_{1} * r}{\gamma_{1} * (EA)_{1} + \gamma_{2} * (EA)_{2}}$$

$$\bar{z} = \frac{\gamma_{1} * (EA)_{1} * \bar{z}_{1} + \gamma_{2} * (EA)_{2} * \bar{z}_{2}}{\gamma_{1} * (EA)_{1} + \gamma_{2} * (EA)_{2} * \bar{z}_{2}}$$

$$r = (h_{1} + h_{2})/2$$



FIGURE 3-1: Two-component composite system

\*h/2 only when centroid of component centered on depth of component



FIGURE 3-3: Mass timber panel on timber beam

	C	Partial Composite Action Factor (γ)					
	Case	Strength & Deflection	Vibration Design				
	Concrete topping on mass timber panel detailed as a TCC system with explicit composite action	From testing or detailed analysis	Potentially higher than for strength & deflection				
Action	Concrete topping cast directly on mass timber floor with nominal connection	N/A <sup>1</sup>	0.15-0.50 <sup>2</sup>				
posite /	Concrete topping cast directly on mass timber floor with no connection	N/A <sup>1</sup>	0.05-0.152				
ntal Con	Concrete topping on acoustic mat or slip-sheet on mass timber panel	N/A <sup>1</sup>	0-0.05 <sup>2</sup>				
Incide	Mass timber panel in direct contact with timber beam with clamping connection	N/A <sup>3</sup>	0.5-1.0 <sup>2</sup>				

TABLE 3-6 Suggested composite action between floor assembly components

<sup>1</sup> Only the mass timber panel is considered; the cementitious topping layer is ignored.

<sup>2</sup> Values are based on limited testing and field observations.

<sup>3</sup> Only the beam is considered; potential contribution from the mass timber panel and topping is ignored.



FIGURE 4-1: Vibration design flow chart

### CLT Handbook Method

- Empirically derived span limit for bare CLT on "rigid" supports (e.g., bearing walls)
- Does not directly apply for CLT panel on beam systems
- Increased span available for multi-span panels with nonstructural elements or cementitious toppings
- Reduced span required for heavy cementitious toppings

$$\begin{split} L_{\lim} &\leq \frac{1}{12.05} \frac{(EI_{eff})^{0.293}}{(\overline{\rho}A)^{0.122}} \text{ (ft)} \\ L_{\lim} &\leq \frac{1}{13.34} \frac{(EI_{eff})^{0.293}}{(w)^{0.122}} \text{ (ft)} \end{split}$$

TABLE 4-4: CLT Handbook method recommended span limit

Grade	tp (in)		G	Density, p, at MC=12% (pcf)	Ρ	A (in²/ft)	
	4.125						12.4
			0.49				
E4	4.125						
E4							
E4		1089					
			0.43	29.7	0.46	49.5	
		"Ru	e ot	217	mb	82.5	
		962	0.43	28.7	0.46	115.5	
	4.125	108	0.49	32.5	0.52	49.5	
	6.87	brati	on S	span	0.2	nts	
	9.625	1027	0.49	32.5	0.52	115.5	
	4.125	108	0.49	32.5	0.52	49.5	
	4 875	/8" 3	-nlv	22.5	10	13ff	
	9.625	1027	0.49	32.5	0.52	115.5	
	4.125	.95	0.42	28.1	0.45	49.5	12.4
	6875	18" 5	-nlv	28.1		<b>1</b> 8ft	
	9.625	898	0.42	28.1	0.45	115.5	
	4.125	95	0.55	36.3		49.5	
	075	/8" 7	-nlv		1 to	22ft	
	9.625	899	-bià	36.5			
	4.125						
							20.4
	4.125						
				41.8			
		114					12.4
	9.625	114					12.4
	9.625 4.125 6.875	114 438		41.8 41.8			12.4 17.3



FIGURE 4-1: Vibration design flow chart



FIGURE 5-4: Implicit vs. explicit modeling of a glulam beam and a mass timber floor panel

Property Modification Factor EA<sub>eff.0</sub> f11 =



\* Consensus on the out-of-plane torsion stiffness of most mass timber products does not exist in the literature. Its property modification factor is therefore conservatively taken as the lower of that for strong or weak axis out-of-plane shear.

#### FIGURE 5-3: Local axes and corresponding stress orientation definitions for one commercially available software platform (SAP2000)

PLATE BENDING AND TWISTING MOMENTS

STRESSES AND MEMBRANE FORCES Stress Sij Has Same Definition as Force Fij

Forces are per unit

Moments are per unit of in-plane length

of in-plane length

Transverse Shear (not shown

Positive transverse shear forces and

stresses acting on positive faces point toward the viewer

#### **Resonant Response Calculation**

$$\begin{aligned} a_{real,h,m} &= \left(\frac{f_h}{f_m}\right)^2 \frac{F_h \mu_{r,m} \mu_{e,m} \rho_{h,m}}{\widehat{m}_m} \frac{A_m}{(A_m^2 + B_m^2)} \\ a_{imag,h,m} &= \left(\frac{f_h}{f_m}\right)^2 \frac{F_h \mu_{r,m} \mu_{e,m} \rho_{h,m}}{\widehat{m}_m} \frac{B_m}{(A_m^2 + B_m^2)} \\ A_m &= 1 - \left(\frac{f_h}{f_m}\right)^2 \\ B_m &= 2\zeta_m \frac{f_h}{f_m} \\ a_{real,h} &= \sum_{m=1}^N a_{real,h,m} \\ a_{imag,h} &= \sum_{m=1}^N a_{imag,h,m} \\ a_h &= \sqrt{a_{real,h}^2 + a_{imag,h}^2} \\ a_p &= \sqrt{\sum_{h=1}^4 a_h^2} = \sqrt{a_1^2 + a_2^2 + a_3^2 + a_4^2} \end{aligned}$$



FIGURE 6-2: Example post-processor output for evaluating floor vibration performance based on resonant response for a given excitation and response node; area in grey indicates walking frequencies of interest



FIGURE 6-3: Example post-processor output for evaluating floor vibration performance based on impulsive response for a given excitation and response node; area in grey indicates walking frequencies of interest



FIGURE 6-4: Example post-processor output for evaluating floor vibration performance based on an envelope of results for all combinations of response and excitation nodes; cooler colors indicate areas of higher (less desirable) floor vibration

f <sub>w,min</sub>	0.8	Hz	h	1		h	2		h	3		h	4	
f <sub>w,max</sub>	2.8	Hz	fh	2	Hz	f <sub>h</sub>	4	Hz	f <sub>h</sub>	6	Hz	f <sub>h</sub>	8	Hz
g	32.2	ft/sec <sup>2</sup>	DLF	0.431		DLF	0.091		DLF	0.071		DLF	0.065	
a <sub>total</sub>	0.01310	g	F <sub>h</sub>	72.3	lb	F <sub>h</sub>	15.4	lb	F <sub>h</sub>	12.0	lb	F <sub>b</sub>	10.9	lb
			N	550		N	1100		N	1650		N	2200	
			Ph	1		ρ <sub>h</sub>	1		ρ <sub>h</sub>	1		ρ <sub>h</sub>	1	
			a <sub>h</sub>	0.000549	g	a <sub>h</sub>	0.000549	g	a <sub>h</sub>	0.0014	g	a <sub>h</sub>	0.013	g

Mode	f <sub>m</sub>	μ <sub>r,m</sub>	μ <sub>e,m</sub>	mm <sup>hat</sup>	Am	Bm	a <sub>real,h,m</sub>	a <sub>imag,h,m</sub>	Am	Bm	a <sub>real,h,m</sub>	a <sub>imag,h,m</sub>	Am	Bm	areal, h.m	a <sub>imag,h,m</sub>	Am	Bm	a <sub>real,h,m</sub>	a <sub>imag,h,m</sub>
-	Hz	ft	ft	lb-ft-sec <sup>2</sup>	-	-	ft/sec <sup>2</sup>	ft/sec <sup>2</sup>	-	-	ft/sec <sup>2</sup>	ft/sec <sup>2</sup>	-	-	ft/sec <sup>2</sup>	ft/sec <sup>2</sup>	-	-	ft/sec <sup>2</sup>	ft/sec <sup>2</sup>
1	6.69	-0.00017	-0.00017	83.33	0.91068	0.01195	2.32E-09	3.04E-11	0.64272	0.02391	2.78E-09	1.04E-10	0.19612	0.03586	1.55E-08	2.84E-09	-0.42912	0.04782	-1.17E-08	1.31E-09
2	7.57	0.00074	0.00074	83.33	0.93016	0.01057	3.53E-08	4.01E-10	0.72066	0.02114	3.87E-08	1.13E-09	0.37148	0.03171	1.31E-07	1.12E-08	-0.11737	0.04228	-5.98E-07	2.15E-07
3	7.99	-0.00044	-0.00044	83.33	0.93740	0.01001	1.13E-08	1.21E-10	0.74962	0.02002	1.20E-08	3.21E-10	0.43664	0.03002	3.61E-08	2.48E-09	-0.00152	0.04003	-2.43E-08	6.40E-07
4	8.00	-0.00576	-0.00576	83.33	0.93750	0.01000	1.92E-06	2.05E-08	0.75000	0.02000	2.04E-06	5.43E-08	0.43749	0.03000	6.11E-06	4.19E-07	-0.00002	0.04000	-4.35E-08	1.09E-04
5	8.00	0.03616	0.03616	83.33	0.93756	0.00999	7.55E-05	8.05E-07	0.75026	0.01999	8.01E-05	2.13E-06	0.43808	0.02998	2.40E-04	1.64E-05	0.00102	0.03998	1.10E-04	4.28E-03
6	8.02	0.09111	0.09111	83.33	0.93779	0.00998	4.78E-04	5.08E-06	0.75117	0.01995	5.06E-04	1.34E-05	0.44012	0.02993	1.51E-03	1.03E-04	0.00467	0.03991	3.13E-03	2.68E-02
7	8.10	0.04037	0.04037	83.33	0.93898	0.00988	9.19E-05	9.67E-07	0.75590	0.01976	9.69E-05	2.53E-06	0.45079	0.02964	2.85E-04	1.87E-05	0.02362	0.03952	2.32E-03	3.89E-03
8	8.11	0.00025	0.00025	83.33	0.93922	0.00986	3.62E-09	3.80E-11	0.75686	0.01972	3.82E-09	9.95E-11	0.45294	0.02959	1.12E-08	7.29E-10	0.02746	0.03945	9.77E-08	1.40E-07
9	8.11	-0.13679	-0.13679	83.33	0.93925	0.00986	1.05E-03	1.10E-05	0.75700	0.01972	1.11E-03	2.88E-05	0.45325	0.02958	3.24E-03	2.11E-04	0.02799	0.03944	2.85E-02	4.02E-02
10	8.13	-0.00244	-0.00244	83.33	0.93951	0.00984	3.33E-07	3.48E-09	0.75805	0.01968	3.50E-07	9.08E-09	0.45562	0.02951	1.02E-06	6.60E-08	0.03222	0.03935	9.40E-06	1.15E-05

FIGURE 6-1: Example post-processor resonant response calculations

## Mass Timber Floor Vibration Design Examples

# Mass Timber Floor Vibration Design Example 1

#### This example demonstrates...

- CLT panels spanning to bearing walls
- Floor system where the CLT Handbook Method is applicable
- Expected performance differences between single span versus multi-span panels
- Modeling capabilities and inputs in RFEM
- Detailed resonant and transient response calculations
- Typical occupancy vibration criteria











#### FIGURE 7-2: CLT floor assembly

FIGURE 7-5: Orthotropic surface properties

FIGURE 7-7: Single-span first three mode shapes

# Mass Timber Floor Vibration Design Example 2

#### This example demonstrates...

- NLT panels spanning to glulam beams
- Floor system where the CLT Handbook Method is not applicable
- Expected performance differences between considering versus neglecting nonstructural exterior walls in model
- Modeling capabilities and inputs in RFEM
- Detailed resonant and transient response calculations
- Typical occupancy vibration criteria



FIGURE 8-4: Mode shapes for mode 1 (left) and mode 2 (right)



FIGURE 8-1: Structural system considered in this example

3" concrete	
1/2" sheathing	
2x10 NLT/DLT	

FIGURE 8-2: Floor build-up

# Mass Timber Floor Vibration Design Example 3

#### This example demonstrates...

- CLT panels spanning to glulam beams
- Floor system where the CLT Handbook Method is not applicable
- Detailed timber-concrete composite action calculations and property modification factors
- Modeling capabilities and inputs in SAP2000
- Detailed transient response calculations
- Sensitive equipment vibration criteria







4 1/8" CLT (V2)

FIGURE 9-1: Floor assembly considered in this example

FIGURE 9-6: Shell property/stiffness modification factors

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https://www.woodworks.org/wp-content/uploads/wood\_solution\_paper-Mass-Timber-Floor-Vibration.pdf

# Thank You

Reid Zimmerman, PE, SE Technical Director | KPFF Portland reid.zimmerman@kpff.com

